

# AVIATION WEEK

JULY 26, 1948

A MCGRAW-HILL PUBLICATION



## Super-Tough Tires

*meet Skyrocket needs*

The Navy-Douglas Skyrocket presents tremendous landing-wheel problems — tires, wheels and brakes must have super-toughness to take terrific take-off and landing speeds. That's why Goodyear extra-high-pressure tires, Goodyear magnesium alloy wheels and Goodyear Single

Disc Brakes are used — they combine extra-ruggedness, longer life and greater wear. For full information about Goodyear Aviation Products, for any type of aircraft, write: Goodyear, Aviation Products Division, Akron 16, Ohio or Los Angeles 54, California.



MORE AIRCRAFT LAND ON GOODYEAR

TIRES TH



HONEYWELL quality is the result of close attention to every engineering detail. The spacing of electrodes in tank areas of the electronic fuel gauge is an example of such careful engineering. Very narrow spacing would have multiplied the design job but the resulting gauge results from condensed attention bridging the gap between narrowly spaced electrodes.

To guarantee the accuracy of Honeywell's capacitor type gauges, Honeywell engineers actually measured the largest droplets of water that could be made to cling to the

electrode surface, then spaced the tubular electrodes .038 inches apart to insure that no film readings would be caused by droplets bridging the gap.

This example of Creative Engineering is typical of Honeywell's refusal to accept anything less than the climate or the quality of Honeywell products. . . A policy that means peak performance, long life, and maximum maintenance cost for the original industry. Minneapolis Honeywell Regulator Co., Minneapolis 5, Minnesota . . . in Canada Toronto 17, Ontario.



In addition to our regular line of Aircraft Instruments for use by Army, Navy, Commercial Airlines and Private Pilots,

## KOLLSMAN

is now producing such special precision devices as:

Barometer (Marine — Sport — Theatre)

Cabin Pressure Control Regulators

Altitude Sensing Indicators

Flight Test Instruments

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Tandem

Mach Meters and Mach Limit Switches

Special Purpose Miniature Motors

Periscopic Sensors

Electric Mechanical Controls for

Altitude, Air Speed and Acceleration

### KOLLSMAN INSTRUMENT DIVISION







## Airlines fly on *Good Groundwork*

- Commercial airline profits are made in the air... and good groundwork in the factories and in the hangars assure more time between overhauls, less time on the ground.
- It's the maintenance man's good groundwork in the hangar that keeps the airline tuned up to take off on schedule... and Sperry's good groundwork in the laboratory, and in the factory helps him do his job more better and quicker.
- Every piece of Sperry's maintenance equipment is engineered and manufactured to give the maximum of trouble-free hours in the air... and for easy accessibility and servicing on the ground at the regular

- scheduled airline overhaul periods.
- To make servicing of Sperry products all easier for the maintenance crew, Sperry conducts special training schools. Here the airline's key maintenance personnel learn to service Sperry equipment and teach others in its best way... all to assure more efficient and more economical operation.
- And in the field, Sperry Service Engineers themselves are always on call wherever, whenever service is

needed. Most of these men are graduate engineers. Their main job is to help the customer by making sure that his Sperry equipment and its installation give him the best possible service.

- Moreover, Sperry research and engineering look to their Field Service Engineers and the many Sperry-trained maintenance men. Among its customers for the vital performance data that help Sperry build new and better air to be aviation



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## NEWS SIDELIGHTS

### Election Prospects

The two top ranking Democrats on Senate Interstate and Foreign Commerce Committee, which handles all civil aviation matters, face stiff primary election fights and are not given odds to win. They are Colorado's Sen. Ed Johnson and Tennessee's Sen. Tom Sawyer.

Johnson is pitted against Eugene Carr, former state Democratic chairman, in Colorado's September 7 primary. Sawyer's chance is the August 5 primary in Tennessee run in tandem because of the holding of "Democrats" from the Democratic rank to the state. Johnson is past chairman of interstate's aviation subcommittee.

Defeat of Johnson and Stewart would mean that Arizona's Sen. Ernest McFarland would become top ranking Democrat on the committee.

### New Senator

Maine's Rep. Margaret Chase Smith, the Republican aviator who appears slated for election to the Senate in Maine's September 14 election, is best known in aviation circles for her leadership in the House fight against the Civil Aeronautics Board.

She has repeatedly charged that the Board favors the interests of scheduled airlines and hinders the development of independent operators. She introduced a resolution calling for an investigation of the matter which died in the Rules Committee. The congressional is well-known figure in aviation circles. Smith often is a member of the old House Naval Affairs Committee and later is a member of the Armed Services Committee.

She was opposed in the primary by Maine's GOP Sen. Owen Brewster, a well-known figure in aviation circles. Brewster supported the strike's G. House leadership for the GOP administration.

### Constitutional Quirks

Apprentice that California's Rep. Carl Albert would not be considered for Secretary for Air in a Republican administration makes a constitutional provision appear to be unconstitutional. The provision states that "no Senator or Representative shall, during the term for which he was elected, be appointed to any civil office... which shall have been created... during such term."

Howland is serving in the 80th Con.

### Arrested Development

Two advanced personal plane design projects in a state of arrested development at Engineering & Research Corp., Bethesda, Md. Fred E. Wink, assistant to the president, says, "I suspect these long before he left E&R to become a professor at Texas A & M College."

Prototypes for both designs, the low-wing 165 hp. Riverside P-100 and the high-wing, two-engine P-101, have been completed for more than a year, but work on them has been stopped due to the declining personal plane market. Both planes are sufficiently advanced in design that they could be tough competitors in one several of the personal plane market in the next several years. The Riverside, a twin-engine, would compete with the Cessna 440, and the P-101, a four-engine, would compete with the Cessna 440, and the P-101, a four-engine, would compete with the Cessna 440.

gens which entered the 1947 Uniformed Air, creating the post of Secretary for Defense and Secretary for Air. Experts report that the constitutional provision means that the congressman could not be appointed from the 80th Congress to serve in defense or air, meaning, but it does not mean that he could not be appointed from a new term of service, in the 81st Congress, to 81 either.

Appointing of Pennsylvania's Sen. Edward Martin as defense secretary in a GOP administration—being widely discussed—does, however, appear to be "out" under the provision. Martin was elected in a November election in 1946, as a member of Congress when the Uniformed Air was passed, and, therefore, experts report, it is not likely to fill part made by that act during the remainder of his term of office.

### National Strike

National Airlines officials are worried at the increasing criticism of management sustained by the President's Emergency Fact-Finding Board's report on the carrier's disputes with the Air Line Pilots Association and with the

### International Association of Mechanics

Authorities on labor relations agree that the language used by the panel was mutually strong, especially the statement that National had exercised "an uncertainty and lack of responsibility inconsistent with the duties imposed by Congress on interstate travel." This warning, plus the finding that NAL had violated the National Labor Act, may be used effectively by ALPA in its appeal for a CAB order ordering the airline to comply with the labor act or suffer repercussions of its certificate.

Other questions on the case is asked told this week. Meanwhile, ALPA has proposed to invite the full support of the American Federation of Labor as an attempt to tie up National's operations completely if the company failed to accept the Presidential Emergency Board's recommendations by the end of last week.

### Transport Problem

If the Russian Maritime on India conference, which for the Air Force to begin using U-2 reconnaissance planes and plans to conduct the shift into the Black Sea. The USSR is already crapping the board on the number of transports available for the operation without curtailing its other operations while some without one have surplus C-54s. Best bet for the USSR is to buy the C-54. C-57. Fuelers already specially equipped for purchasing heavy equipment and supplies. Most of the 280 Russian declared to the USAF are tied up for the summer in heavy winter training maneuvers.

### Training Revolution

Fourth Fighter Wing, Strategic Air Command, Andrews Air Force Base, Md. has recently named the Air Force to do some out-of-the-box in its highly-located scientific methods of component pilot in training. The Fourth has recently placed AT-6 pilots up to the seat of a F-86 fighter and watched the acrobatics handle the jet fighter like a veteran in a few hours practice during recent weeks.

An Training Command is proceeding steadily with its Fall syllabus which will place AT-6-trained pilots in F-86C training after 170 hours. The this advanced system was introduced only after extensive acrobatic study, the Fourth jet tried it as a task, and used the taxpayer almost thousand dollars on each Air Force pilot.





## SHAKING THE TRUTH OUT OF A TURBINE BLADE

- Here a pair of electro magnets is "shaking the truth out of a turbine blade." It is being shaken in this laboratory to determine its true natural vibration frequency—the dangerous frequency that exists when a very small force causes a large deflection.
- The blade is actuated by high frequency magnetic impulses. As the speed of the magnet continues to increase, the blade is made to vibrate at its various natural frequencies. A photo-electric cell serves to locate these frequency precisely, while a

measuring microscope reveals the exact amplitude of each.

- Modern research such as this takes more blade shaking in a much shorter time than would otherwise be possible and provides accurate data from which engineers can design turbine blades and many other vital aircraft engine parts that do not possess harmful vibration characteristics.

- Another example of the painstaking research behind the development of Wright aircraft turbine and reciprocating engines.



POWER FOR AIR PROGRESS

# WRIGHT

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WORLD'S LARGEST WIND TUNNEL and latest wing plus jet and rocket powered supersonic interceptors are shown here.

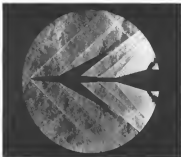
## NACA Shows Ames Laboratory Progress

**New data is offered on the supersonic roles of swept, straight wings.**

By Robert Blom

Moffett Field, Calif.—Significant advances in shaping the details of supersonic aerodynamics plus new transonic research techniques were revealed in the second biennial inspection of the Ames Research Laboratory at the National Advisory Committee for Aeronautics. More than 500 aircraft manufacturing executives and engineers plus top ranking Air Force and Naval Aviation brass attended the two-day session.

A more distinct, detailed picture of the aerodynamic phenomena encountered in the range between Mach 3 and Mach 2.0 has emerged as a result of the last year's NACA research in this previously undefined area. Most important advance of the year—the first planned supersonic flight by USAF and NACA test pilots in the Bell X-1 rocket-powered smooth-plate-mat supersonic



omitted from one of the dimensions.

► **Sleep in Straight**—Perhaps the most important dimension offered by the NASA researchers was the argument of proper places to avoid lock and straight wings in the supersonic picture. The supersonic flights of the X-33 program it is possible to achieve these goals with extremely thin, straight wings. But wind tunnel research and flight test data indicate that the swept back wing still offers the most promising modification in the transition range between Mach 0.8 and about Mach 1.8.

In this case the sweep back of the wing delays the formation of shock waves on the wing surfaces, a welcome side effect over the wing while the aircraft is travelling at supersonic speeds (above photo, p. 11).

At about Mach 1.8 the shock waves catch up with the swept back wing and end up drag making leaving the capabilities of any power plant that can likely be available in the foreseeable future. A straight wing encounter shock waves below Mach 1.6. Above Mach 1.8 the straight wing becomes more difficult than the swept back wing.

► **Thermal implications at this:**

► **Heatstroke** long range, heavy load carrying planes such as bomber and transport operating in low supersonic regimes will feature swept back wings. ► **Ultra-sonic** and intercept types will have configurations much like the B-2, but with swept wings and engines in transonic power which will probably be available only for extremely short periods.

► **New Technology**—New supersonic research techniques involving the use of the shock wave device of Schlieren photography of shock waves to transfer photos made in actual flight tests on high speed aircraft. This differs from previous "frozen snapshot" photography of shock waves in flight in strong sunlight because it gives a detailed view in a virtual plane through the shock wave and secondary layer. Schlieren photography actually shows the shock wave boundaries by using the light.

The Schlieren apparatus is attached to an F15 wing in shock tunnel studies. A train of parallel light rays is projected through a narrow slit, across the wing. Double vanishing reflect the light and register the shock wave and other air density changes on the camera lens.

► **Diffusionless**—Principal difficulties encountered in aerodynamic analysis of this winged and transonic flight tests were interference of dust and reflected sunlight and plane vibrations.

Sight interference was overcome by the use of a light shield of angle the agency pointed with a camera filter that admits light only at the frequency of

the projecting beam. This excludes stray light of other frequencies from register in the film. Use of shock waves (shockwaves) that disturbed the optical system.

► **Transition**—A required item for the pilot in this type of flight test to keep an excellent check, were on his wing within the three inch field of the Schlieren apparatus.

Another critical supersonic problem requiring early solution is safe pilot entry into extremely high speeds. NASA has been testing permeable nose which forces the pilot down into intake speeds before using a parachute.

► **Transition** of these regimes to turbid and over and over after parting from the plane produces leads up to 20 Gs, considerably more than a pilot can stand.

Need for special stabilizing devices to prevent tumbling, avoid twisting of the sort of the plane, and slowing the engine to speeds where a chute can be used safely is anticipated. Wind tunnel tests have resulted in frequent collisions between the engine and planes that are designed to be of control.

Among the other field of supersonic research discussed:

► **Heat Transfer**—Discussion of temperature test given by six factors at ultimate speeds (area of many problems). Indications are that refrigeration systems will be required for aircraft operating at supersonic speeds to high as 10,000 ft. where extremely cold air is encountered in service. The formation of as high as 2300 degrees Fahrenheit will be associated between the front and rear portions of the aircraft involving at 75 ft. miles altitude. Research has also revealed that drag dominates at least by several thousand times thermal strain during prolonged tests.

► **Control**—For planes that have shifted from trailing edge boundary layer control devices to a leading edge suction slot or porous strips as a means of restoring lift and improving stall characteristics of extremely thin models required for supersonic flight. Power requirements to produce the surface suction to be large and will require considerable reduction before the method becomes practicable.

► **Propulsion**—Research is continuing on propellers up to supersonic speeds with sweptback and thrust blades the most promising forms under consideration. Areas of 50 sq ft. wind tunnel is being used to test influence of fueling and wing efficiency of this engine aircraft on propeller efficiency.

Those attending the symposium passed to give a scientist's short lecture to the University of Virginia, where discussion of NASA research for 25 years, whose death was maintained in the perspective.

## Air Legislation

The special session of Congress may act on left-over measures.

Congress is expected to complete action during the special session coming today on important aviation legislation which must pass in the last session each year for adjustment a month ago.

The session, all of which have overwhelming support in Capitol Hill, are:

► **Preemptive Development**—Legislation authorizing the government to finance research and development work on commercial cargo and transport planes was blocked from unanimous consent passage as the closing hour of the last session of the Senate by Sen. John Williams (R., Del.) and Sen. Claude Pepper (D., Fla.). Reconsidered by the Congressional Aviation Policy Board and all concerned agencies concerned, it already has passed the House and has the backing of a majority in the Senate. Sen. Owen Roemer (R., Mass.) plans to re-introduce it through during the special session.

The program would be directed by a five-member government board of representatives of the Air Force, Navy, Civil Aeronautics Board, Civil Aeronautics Administration, and National Advisory Committee for Aeronautics, with the assistance of a seven-member industry advisory board. Funds for the program would come from Air Force appropriations and the Secretary for Air would have veto power over proposed projects.

► **Five-Year Procurement**—Legislation authorizing the very program of development projects was approved by the Senate but did not reach the House Armed Services Committee. It provided that National Defense Intelligence research and development funds allocated during the year for which they were made, that remain available for re-allocations or expenditure for the four following business years.

It also authorizes the Secretary for Air to engage an independent number of scientific advisors (at \$10 a day, plus \$100 subsistence), and to furnish up to \$100,000 (the cost of research, development, or test facilities required by contractors).

► **National Security Foundation**—Legislation already passed by the Senate and approved by the House, authorizes the National Security Foundation to assist for House clearance only in the Senate. The Foundation, directed to promote and coordinate basic re-

search as national defense and other scientific fields, would be composed of 25 members appointed by the President with the approval of the Senate and headed by a director also appointed by the President with the Senate's approval.

► **McGuire-Hill**—Legislation authorizing numerous authorized strength for USAF was passed by the House, but was not taken up by Senate Armed Services Committee because of lack of time. The bill sponsored by Rep. Charles Stenholm (R., Minn.) provides for 78 regular Air Force groups and 25 separate squadrons, supplemented by 27 National Guard groups, 18 Air Reserve groups, and auxiliary units, total personnel strength of 503,000, plus civilian personnel "as may be deemed necessary", a plane force of 34,800, or 22,500 aircraft; no more than 10,000 aircraft on military bases at 5200 planes, or 42,000 national time.

► **Profit Limitation**—Legislation repealing two unrelated provisions of the recent Internal Anti-theft Statute of 12 aircraft parts on military and naval aircraft and the requirement that 10 percent of the Navy's planes be manufactured by the government-owned Philadelphia plant will probably be brought to a vote and passed by the Senate.

► **NACA Personnel**—Legislation authorizing NACA to have two additional \$1,500,000 professional and scientific personnel, approved by the House and the Senate Post Office and Civil Service Committee, is scheduled for early action by the Senate.

► **Traffic Protection**—Three comparative measures on measures to protect aircraft and promote international travel, introduced by the Congressional Aviation Policy Board, will be pushed in the special session by Bowser.

The would require the requirement that aircraft pay the cost for customs inspection services on Sundays, holidays, and at night. Passed by the Senate, the measure is pending before the House Ways and Means Committee.

Another would fix an aircraft ticket fee for transporting as much as the U.S. 4 of the line had a vote raised by the U.S. committee after debate. Cases would be required only to leave the cost of maintenance of the plane while on land, and return fees to the country of its origin.

The three measures, sponsored by Bowser, would give the President authority to execute an agreement with the governments of a sovereign country under which nationals of the two countries (for the preceding 16 years) would be permitted to travel between the two countries as passengers in the other country without obtaining visas—a complicated, time-consuming process.



ANOTHER RUNWAY goes down at Tempelhof. This, like the one in use, is paved steel plate, and will accommodate C-47s and C-54s on the Berlin run.

## Report From Berlin...

(The greatest assistance in supply operation over underground is keeping alive about 1,500,000 people in the Western sectors of Berlin. Last week, McGuire-Hill would have been passed by the House for the bill for the following report from the Congressional Committee.)

► **BERLIN**—"Operation Vulture" is a military action of the Berlin Air Force to build their own supply line to the Western sectors of Berlin, the "West power" Air Supply Center clears the pilot for a ground landing to the field traffic control zone.

► **Operations**—The U.S. has been operating two groups of C-47s, and one of C-54s. The groups very in size and in the size of the C-54s only 28 of the present 32 C-54s have been taking to the air at one time.

Plans have been taking off as an average of four minutes, although with a 20-minute interval between the last plane of a C-47 group and the first of a C-54 series to compensate for the C-54's greater speed.

► **Weather**—Early estimates, particularly conditions, the groups have been flying those outside daily. A maximum average eight hours, including time in the ground, the loading, unloading, and fueling, and gatekeeping. Civilian work two night-hour shifts in one recent day, 201 U.S. planes landed at Tempelhof Airfield and 204 British planes at the Gatow Airfield.

► **Conditions**—There are three 20 mile wide air corridors into Berlin. One is used only by U.S. Berlin-bound planes. Another is used by both American and outgoing British planes, and by German U.S. planes. The third lane is used in both directions by the Scandinavians.

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## Pilots Call For Cockpit Changes

**IAS meeting hears recommendations on high-speed planes. Haze forecasts jet transports coming by 1951.**

Los Angeles—Demands for radical changes in the cockpit design of super sonic planes and a prediction that commercial jet transport will become profitable by 1951 were reflected in comments at the annual midwinter conference of the Institute of Aeronautical Engineers here July 15 and 16.

Three Douglas test pilots called for more pilot locations in extremely high and very high altitude cockpits. Robert E. Hagen, Boeing Aircraft Co. described a jet car in which "should provide a complete service and be capable of landing, 100 miles per hour distances in hours."

Such a plane, he said, would carry 30 passengers, cruise 500 mph at 15,000 ft., and operate at 10,000, 20,000 and 25,000 to 30,000 miles, and "has to be technologically feasible and commercially profitable by 1951." He is convinced that public demand will force solution of altitude and increased loading problems notwithstanding jet transport operation.

Mostly Miltner—Hagen's paper was the sole exception to the usually solitary flow of engineering reports and presentations. For the first time since the IAS required the signing of a certificate of opinion for admission to its sessions, a pilot's experience on high speed flight. There was a dearth of discussion of engineering problems concerning civil aircraft now in operation.

Air Force Secretary W. Stuart Symington outlined progress of U. S. aviation achievements and spoke at length in support of the 70-Group Air Force program.

Rebuttal resolutions covering

aircraft and aerodynamics at high Mach numbers, developments at Lockheed's battery of supersonic wind tunnels at Santa Barbara and a simulated glimpse of activities at Navy's Air Missile Test Center at Ft. Worth near Los Angeles.

The test pilots from Douglas Aircraft Co. were Gene May, Robert French and Russell W. Thow. For several flying up to Mach 2.0, they demanded assistance of the pilot in a fully pressurized cockpit and, not pilot freedom from past environments.

Flight Equipment—Among those they listed were helmet, oxygen mask, oxygen canopy bottle and regulator, "G" seat, parachute, life preserver, radio master battery, goggles, gloves, boots, pressure suit, and other "gadgets."

They stated that design engineers

give serious attention to production of a jettable cockpit to equip that the pilot's communications will be limited to "C" seat, not microphone and setting out.

He recommended, as a step toward the fully integrated cockpit, that high speed, high altitude aircraft now being developed employ automatic air brakes to slow the cabin to a specific air speed, around 500 mph, at the pilot through the cabin floor to reduce oxygen shock hazard, and in control cockpit pressure differential in a cabin pressure of not less than 12,000 ft. with the airplane at 40,000 ft.

The pilots and Dr. C. F. Lanchester, uncommitted research authority of University of Southern California, wanted to tell them with the proper design of specialized cockpit and escape device the chances are good for survival of a flyer encountering structural failure at very high speeds and altitudes.

Dr. Lanchester declared that fast-flight escape from altitudes considerably above

40,000 ft. should be possible. He said, "There are some indications that a pilot might survive a free-fall escape from as high as 60,000 ft."

A major problem of pilots at high speeds, Dr. Lanchester said, is "psychological stress." He urged that in the interests of flight safety new altitudes be given to making the pilot comfortable in his aircraft and in reducing the complexity of his flying duties.

## Engine Procurement To Top Half Billion

Air France and Naval Aviation will buy \$517,000,000 worth of aircraft engines for fiscal year 1949. Official Air Force announcement of contracts follows by a foreign American. West's July 15 report outlining the engine procurement program for aircraft ordered during fiscal 1949.

Vastness of engine delivery schedule based on individual company problems has forced Air Force to sever drastically its engine procurement program to justify one for the defense program. The engine has deferred portions of the engine program for fiscal 1949 and 1950 and accelerated other portions for rapid delivery during fiscal 1949. These actions, however, do not change the overall engine requirements for aircraft which are ordered during fiscal 1949.

Largest volume, both now and for the future, is first contracted by both services with Allison Division, General Motors Corp. which will build 2071 Air Force and Navy model R-3350 engines, 1000 hp, for the Navy and the Air Force. The R-3350 engine is power the Lockheed F-80, Grumman F7F and Martin P-4M aircraft. Air Force will also receive 1125 Allison P-35 non-flow engine for the Navy. The R-3350 engine is power the Lockheed F-80, Grumman F7F and Martin P-4M aircraft. Air Force will also receive 1125 Allison P-35 non-flow engine for the Navy. The R-3350 engine is power the Lockheed F-80, Grumman F7F and Martin P-4M aircraft.

Largest propeller engine award will go to the Pratt & Whitney R-4350 V-12 engine, 3000 hp, for the Navy. The R-4350 engine is power the Lockheed F-80, Grumman F7F and Martin P-4M aircraft. Air Force will also receive 1125 Allison P-35 non-flow engine for the Navy. The R-3350 engine is power the Lockheed F-80, Grumman F7F and Martin P-4M aircraft.

The "NAV panel" includes one representative each from the Departments of Commerce, Navy, Air Force, Treasury and from the Federal Government Commission and will provide engineering of the research and development phase of the project. After coordination of the functional requirements and operating procedures for a joint military, naval, and air force control system, the panel will submit its recommendations to the ANRDS, which has final say on the form of the program.

## ECA Aid to France

France will buy \$965,000 worth of U. S. aircraft equipment for its commercial airline.

Procurement of this equipment was authorized last week by the Economic Cooperation Administration and will be funded through normal commercial channels. This was the first machine purchase to be approved under ECA and indicates that further approvals will be made for participating countries whose aircraft are American type equipment.

The French purchase will be confined to engines, propellers and spare parts for Douglas DC-7 and DC-4 and Lockheed Constellation type transports now operated by Air France. French authorities pointed out that the ECA purchase will permit only maintenance of current American equipment used in France and will not be sufficient for expansion of the French civil air fleet by purchase of new American planes.

Paul G. Hoffman, ECA administrator, cited record crop crops throughout the world this year as an indication that more ECA funds would be diverted from relief food needs to capital equipment and machinery maintenance items.

## Financial World Awards

On the basis of annual performance reports, 11 aircraft manufacturers have qualified for "Highest Merit Award" citations from Financial World Magazine, and are now candidates for the award as "best of the aircraft manufacturing industry."

Last year's winner was Cessna L. Martin Co., which had the best 1946 annual report of the aircraft manufacturers. This year's winner will be announced at the Financial World annual report awards banquet Oct. 23, 1948, at the Hotel Pennsylvania, New York City.

Other top "Highest Merit Award" mentions: Boeing, Cessna, Douglas, Fairchild, Grumman, Lockheed, Glenn Martin, McDonnell, North American, Republic, Ryan, Solar and United Aircraft.

## AA Raises Capitol Pay

American Airlines has concluded negotiations with its pilots on a new minimum minimum wage scales. New pay scale approved by ALPA, headquarters, the new agreement would set the capitol members from \$238 per month to \$290, the maximum from \$410 per month to \$530.

The minimum-rate increases to maximum will be obtained in twelve stages, each stage six months long. When approved, the pay increase will be retroactive to May 1, 1948.



Art's conception of North American T-28.

## New Trainers for the Air Force

**Order for 268 T-28s to North American; development contract for novel single-engine type to Douglas.**

Awards of a production order for 268 T-28s to North American Aviation, Inc., and a \$552,000 Air Force contract for trainer development to Douglas Aircraft Co. are the two highlights of a rapidly accelerating trainer program under USAF expansion plans.

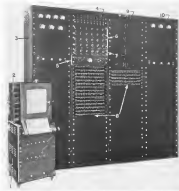
The North American T-28, winner of an industry wide competition for a light purpose AT training plane to replace the old biplane trainer, is a conventional in design except for its tractor propeller.

The two place craft utilizes a Wright R-1300 engine of 600 hp and has top speed of 132 mph, with a cruise of 11,650 ft. It has a normal gross weight of 2862 lbs.

The Douglas contract calls for design of a mockup







Components of electronic analog computer: 1, power supply; 2, amplifier; 3, integrator; 4, summing amplifier; 5, switch; 6, scale factor potentiometer; 7, control potentiometer; 8, scale factor potentiometer; 9, control potentiometer; 10, power supply for the computer.

## Office-Size Electronic Computer

New device, born of war development, provides fast solutions to complex equations used in research.

By Herbert H. Adams\*

Solution of complex mathematical equations by hand method has always been one of the least time-consuming factors in aircraft development projects.

Though electronic "brain" were constructed before and during the war to supply solution of involved mathematical problems, greatest drawback to their general application was their tremendous size and cost.

Now available is an office-size version of a computer originally developed for the Special Devices Center of the Office of Naval Research, and also used by the National Advisory Committee for Aeronautics and major aircraft companies.

Cost, Time Cut—The refined version, marketed commercially by Texas Instruments Corp. as REAC (Reactive Electronic Analog Computer), is a

table-top electronic computer capable of solving higher order, ordinary, linear and non-linear differential equations.

Using the computer in problems substantiated by various airplane companies and universities, it was found that cost of obtaining answers as adequate as those resulting from hand computation was reduced by as much as 95 percent. Problems which were entered to require 2948 man-hours for solution by trained mathematicians using mechanical methods were solved in 108 minutes using the computer.

Cost of hand solution was \$73,725 compared to \$7246 for computer solution.

Components—The complete device consists of four cabinets, each requiring about as much floor space as a filing cabinet, and a secondary unit. Only the recorder unit and two cabinets—computer and control cabinet (Fig. 1)—need be accessible to the operator.

The other two cabinets—power supplies and read by the operator—may be installed elsewhere.

In the computer unit are summing, integrating and averaging amplifiers, scale factor potentiometers, and initial condition potentiometers.

In the accessories unit are the recorders and solvers.

**Amplifier—Summing amplifier** is a d.c. device which has the property of forwarding its output voltage having a negative value equal to the algebraic sum of several voltages applied at the amplifier input. As many as seven input voltages may be applied to such summing amplifier. Each input has a separate amplification factor, and has an amplification of ten, two have an amplification of five, and four have an amplification of one.

**Integrating amplifier** is a d.c. unit provided with a feedback capacitor. It provides an output voltage, which is both negative and equal to the time integral of the algebraic sum of the input signals.

**Integrator amplifier**, whose principle of operation is identical with that of the summing amplifier, is used to change the sign of a variable both in summing amplifier is provided with two inputs so that they may be used as summing amplifier if necessary.

There is a total of twenty amplifiers in the computing unit—seven integrating, seven summing, and six inverting amplifiers. Each is mounted on a separate chassis. If an amplifier fails and it is desired to replace it without writing for repairs and trouble shooting, it may be removed from the chassis, like a plug from a wall socket, and a new unit substituted.

**Potentiometers**—In addition to the amplifiers, there are 24 scale factor potentiometers, controlled by knobs on the front panel. These are provided to get other than unit coefficients of 10, 4, or 1. When a scale factor potentiometer is connected to the input of a summing or integrating amplifier, the amplification factor may be varied to any fractional value.

Also included in the computer unit are six initial condition potentiometers, mounted on the amplifier chassis with voltage equal to the constant of integration.

Inputs and outputs to each amplifier and potentiometer are connected to jacks in the front leg mounted on the front panel of the computer. The basic units involved in a problem are then connected by means of patch cords.

The computer unit can be used without the accessories unit in solving ordinary linear differential equations with constant coefficients up to and including the seventh order.

**Problem Panel**—Typical of the prob-



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### ON THE GROUND

DESIGN, DEVELOPMENT, MANUFACTURING, MAINTENANCE, REPAIR, OPERATION, AND AIRCRAFT PERFORMANCE INFORMATION

laws which fall into this category is that of the determination of the longitudinal stability of a transport aircraft having a proportional type of automatic pilot. In this pilot system, the elevator is automatically operated so as to compensate for any deviation in path from the horizontal, as produced by gusts and the like.

The problem is to determine the proportionality between deviation in path and the resultant elevator deflection, within the time delay of the control system, so as to produce the specified flight. The equations of motion in open loop form are:

$$(p + 2\zeta\omega_n - p^2 + \omega_n^2) = 0 \quad (1)$$

$$(p^2 + 2\zeta\omega_n p + \omega_n^2 + 90L) = 0 \quad (2)$$

$$L = K(1 + \rho p) \quad (3)$$

where  $\rho$  is the proportional operator ( $d/dt$ ),  $\omega_n$  is angle of attack,  $\zeta$  is pitch angle,  $L$  is elevator deflection, and  $K$  is the proportionality constant.

Initially, first  $\omega_n$  at computer time equals zero, all variables are zero. A disturbance such as a gust can be introduced as the derivative of a step function in voltage corresponding to angle of attack. A block diagram of the corresponding computer circuit is shown in Fig. 2.

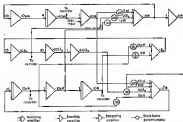
All inputs are disconnected by flipping a single switch while the problem is being set up and the step function inserted. When the circuit has been completed, a switch is thrown which sets the inputs in the appropriate manner, and the solution is recorded by picking off the voltages at the outputs of the appropriate amplifiers as indicated in the diagram.

**Solution Required:** The recorder is a speed in channel oscillograph which makes a permanent, directly readable record of the solution. When the solution is completed, the starting switch is thrown back, opening the inputs again, and resetting the initial conditions.

Merely changing the appropriate potentiometer setting between solutions provides an extremely rapid comparison of the effect of proportionality ratio on stability. For this particular auto pilot problem, approximately two runs would be required to prepare the equations, with three runs for the device and obtain and process 75 solutions. Time required for numerical solution would run into months.

If the particular aircraft were still in design or flight test stages, it is hardly simple to determine the effect of design in configuration, aerodynamic characteristics, and the like, without extensive wind-tunnel coefficients, or possibility of endangering a pilot's life in designing the aircraft.

**Nonmechanical Unit:** Additional range and flexibility is added to the machine by the use of the servomechanism unit. This contains some which are



Schematic of circuit for assumed position of stability of aircraft with automatic pilot.

used to vary out multipliers and division of two numbers, and involves which provide trigonometric functions of a variable.

This unit makes it possible to solve linear differential equations with variable coefficients, as well as many types of non-linear differential equations. A feature of the servomechanism unit is that a special functional potentiometer can be provided, which when connected to a zero shift, will represent any reasonable functional relationship between two variables.

In the auto pilot problem, for example, if the airplane were supercruise, the servomechanism unit would permit an instantaneous change of the elevator and flight path during acceleration from subsonic to supersonic flight. The change in aerodynamic derivatives with Mach number, the change in total forces and moments with velocity and density variations to expose forest characteristics, and other variations could all be taken into account.

In addition to these problems, the machine may be applied to many other problems of the same class. In the field of mechanical design this would include studies of mechanical parts.

For jet, turbo-prop, gas turbine, jet and rocket engines, the machine may be applied to many other problems of the same class. In the field of mechanical design this would include studies of mechanical parts.

The machine may be applied to many other problems of the same class. In the field of mechanical design this would include studies of mechanical parts.

In these applications, the equations of motion, thermodynamics, kinematics and the like are punched into the machine and performance is observed as

der various combinations of system parameters and initial conditions. The range of conditions which solutions are obtained and the ease with which additional features may be added or introduced permits a thorough investigation before design is undertaken.

**Structure:** Applied-Flow-Mechanics problems in aircraft structure can also readily be applied to solution. These include the deflection of skin and internal structure under varying air loads and surface loading conditions, and stress in landing gear, both in operation during landing or take-off, etc. Records of solutions may be used to accurately establish the modulus of velocity acceleration diagrams, and the structure effects.

In addition to pure simulation, the computer may be used in a solving device. For example, referring to the problem illustrated in Fig. 2, instead of simulating the automatic pilot by means of equation (1), the actual auto pilot may be tied in to the machine in such a manner that a d.c. voltage proportional to elevator deflection is fed into the machine.

The machine may have only the two equations of servomechanism patched in, and therefore produces a voltage proportional to the pitch rate. This is used to drive a pitch rate by a rate feedback loop, which is used as a rate feedback loop. The auto pilot mounted on this while needs exactly as under flight conditions. Engine and propeller control systems may be tested in this manner.

The REAC, as dropped and constructed by Harry D. Black, chief engineer and general manager, and Ray J. D. McElroy, project engineer. Techniques of applying the equipment to various problems in aircraft design were developed by Elmer Shorling and Stanley Fink, heads of the analysis and computer group, in the course of eight months' work.



## XS-1: Design and Development

**World's first supersonic plane outgrowth of meeting held in 1944. Although of conventional configuration, craft has novel features.**

By Robert McManis

In the dawn of high speed flight, there is little doubt that attainment of supersonic speed by the Bell XS-1 research airplane on Oct. 14, 1947 marks an historic experience with the first flight of the Wright 3-jet on Dec. 17, 1901.

Although it required slightly less than 44 years for man to fly faster than sound, there exists little mystery as to how it was accomplished; the aeronautical engineering has had in its own objectives throughout all these years, two major factors that made the achievement possible—high power and low drag.

Because the fastest airplane, other things being equal, is the one with highest power and lowest drag, these two factors determined the basic lay-

out of the XS-1 right from the start.

Since speed was its fundamental requirement, considerations of maneuver, combat component, range, maneuverability, serviceability and cost did not compromise the plan, hence the task was simplified.

**Program Formulated**—The need for a special research plane capable of supersonic speed had become increasingly apparent to government agencies and industry in the fall of 1945 as wind tunnel work pushed up into the transonic speed regime and difficulties associated therewith became pronounced.

Representatives of the National Advisory Committee for Aeronautics discussed their problems informally with engineers of the (then) Army Air Forces and Navy Bureau of Aeronautics, who readily saw the potential benefits to be

obtained from related-aerodynamic research at speeds near that of sound.

These informal discussions precipitated a joint NACA-Air Force-Navy meeting at Langley Field, Va., in March, 1944, during which a broad program of high speed piloted flight research was formulated.

The joint program authorized the NACA to prepare basic design data for a series of aircraft capable of flying faster than the speed of sound, to design and develop special research instrumentation for these craft and to act as expert advisor to the services on the project.

The Air Force and the Navy were obligated to provide the funds, administer the procurement of the aircraft, provide the test site, equipment and supporting facilities, and to deter-

mine the technically useful portions of data obtained.

**Bell, Douglas, Bellco**—NACA design studies began immediately, and wind tunnel tests, performance calculations and preliminary requirements of the aircraft had crystallized by December, 1944, at which time Bell Aircraft Corp. and Douglas Aircraft Co., Inc., were invited to participate in the program, designing and building the research plane.

The Bell project was designated the XS-1 and was assumed by the Air Force. The Douglas project was designated D-570 and was assumed by Bellco. Both projects were later expanded to include sweeping configurations.

**XS-1 Requirements**—The Bell XS-1 contract was let in the form of a request for Air Force procurement contract specifying in detail the aircraft requirements to be met and features to be incorporated. It was decided that the XS-1 should, first of all, be a conventional design in order to determine if the transfer configuration could attain supersonic speed.

Experiments were to employ only readily available design techniques such as would be expected in the quantity procurement of a military plane. Only wind tunnel requirements, apparent from available research information, was extremely high structural strength, but again that of any previous design.

Other contract requirements included a test equipment load of 500 lb., deceleration in flight of an 8G pull-out at an indicated speed of less than 500 mph, an 8G pull-out at maximum speed, proof of the speedometer at rated thrust, takeoff and climb to 35,000 ft. under its own power, and satisfactory response to controls at a speed of Mach 0.8.

**Motor Plan**—The end of the war in August, 1945, made available for the project a Boeing B-29 bomber, and it was decided that the airframe method would be used for the flight test program, thereby saving room thus held the available fuel and reducing the considerable inherent hazards involved in an airplane with a wing loading of approximately 100 lb. per sq. ft. (The XS-1 has not made a ground takeoff to date, nor are there any plans for such an attempt with the present airplane.)

**Glide Tests**—The craft was completed late in 1945 but the rocket engine was not available, and plans went forward for glide tests without the engine. During February, 1946, 10 glide flights were made by the late Jack Woolson, at Ft. Pierce, Florida. The Air Force 6-47 had 15,000 ft. runways, although not more than 5000 ft. was ever used in an XS-1 landing.

These glide tests began at about



Narrow confines of pilot cabin are evident in this view of XS-1 being suited for flight.

Unusual cage-like and loading gear arrangements are apparent in this rear view.



27,000 ft and the XS-1 touched down at about 130 mi in a span of about 110 mph. The plane was flown in the safety condition, the purpose being to perfect the landing technique.

During the tests it was determined, as predicted, that the air-scooped method was entirely safe and satisfactory regardless of the speed, power or flap setting of the B-29 mother plane because of a positive pressure region between the two airplanes created by mutual aerodynamic interaction. This pressure provided a natural "push" to the XS-1 rendering unnecessary the special guide rails and vector unit designed for the purpose.

The tests also determined various stability and control of the design.

► **The Motor**—The rocket engine was delivered, and special equipment for rocket fuel storage and handling erected at Muroc Air Force Base, Calif.

This consisted of a 15,000-gal storage tank for liquid oxygen, 1100-gal storage tank for liquid nitrogen, and a special mixing tank for alcohol, and other materials on a small fuel truck.

A jet was dug into which the XS-1 could be rolled for attachment to the B-29.

The engine was installed and ground tests conducted at the Bell plant from June to September. Following this, the plane was flown to Muroc in the B-29 early in the fall of '46. Four glide tests were made with increasing gross weight, by Colonel H. "Mac" Glines, who after "Woolsey's" death, had been assigned as pilot for the contractor's demomstration flight program. On the last of these glide tests both the design test load was met and attained satisfactorily.

► **Powered Flight**—Finally, on Dec. 9, 1946, the XS-1 was carried aloft and

hoisted at 27,000 ft. Ten seconds later No. 1 rocket chamber was fired. Achieved shortly by No. 2. The unit received as rapidly, however, that No. 2 chamber was cut out and the plane moved up to 55,000 ft where No. 2 chamber was again ignited.

With this 50-second power, the XS-1 moved quickly to Mach 0.79, when all power was shut off. The plane glided down to about 15,000 ft and all four chambers were cut in. This last application of full power resulted in an extremely rapid climb and all motors were cut off.

The XS-1 glided to a landing after its first "dash" during which a speed of 550 mph was attained, radio-therm, position sensor control and handling characteristics demonstrated, and a clear indication of the future possibilities of the design obtained.

Bell pilots completed 20 powered flights demonstrating the superior performance of the XS-1 to the Air Force, and only in August, 1947, Capt. Charles E. Yeager was assigned to the project. Meanwhile, the second airplane had been delivered to the NACA, and research pilots Howard G. Lilly and Herbert H. Hoover assigned to the project.

► **Some Speed Recorded**—Powered flights continued at increasing speeds altitudes and rates of climb until Capt. Yeager exceeded the speed of sound for the first time in history—on Oct. 14, 1947.

Shortly thereafter, NACA pilots Lilly and Hoover and Air Force Maj. G. F. Lewis and Capt. J. T. Finn-Gerald exceeded Mach 1.5 in the XS-1 flights. To date, about 55 flights have been made in the two XS-1s, and in about a dozen of these flights supersonic speed was attained.

► **Wing Details**—Detailed design of the XS-1 was completed by Robert J. Wood, Bell chief design engineer, Stanley Smith, original project engineer, and Richard Furr, present XS-1 project engineer.

The wing planform as NACA #5110 section on the airplane being flown by NACA, and as NACA #5108 section on the Air Force craft. This profile is one of a series of bending airfoils developed by the NACA during the war. It has a design lift coefficient of 0.1 and the maximum pressure point is at 50 percent of the chord. It has a critical Mach number of about 0.85 and a low drag range of lift coefficient from slightly more than 0.85 to slightly less than 0.15, which are obtained at an angle of attack of about 1.0 deg. At angle of incidence of 2.0 deg, it is still on the X-8.

The wing has a span of 28 ft., area of 156 sq ft., aspect ratio of 6, and equal taper on leading and trailing edges. It is, obviously, a conventional wing in layout.

To meet the design load factor of 8.0 specified by the contract, the skin is machined tapered from slightly more than 0.5 in. at the root to slightly more than 0.125 in. at the tip. Thin curved skin thickness supplied both the design and fabrication problems and little difficulty was experienced with providing local stiffening and preserving the profile surface to remain highly accurate both.

The port wing panel has 240 drilled holes with special inserts comprising pressure relief for distribution measurement in flight.

► **Stabilizer, Control Cable**—The stabilizer is of similar design and construction to that of the wing but is fitted with a control operated by a pusher member. The push has a high-pitch thread permitting extremely rapid adjustment of the stabilizer incidence, angle to permit it to accommodate large changes in downwash loads quickly.

Rudder and elevator control cables are carried along the top of the fuselage in a special fitting to prevent interference with the spherical joint tanks within the fuselage.

► **Fuselage Cabin Details**—The fuselage is of conventional monocoque construction at 145 ft.

The pilot is located in the nose, behind transparent panels that conform to the upper shape of the forward fuselage.

These panels were double panes of Plexiglas, the area between the panes being delineated in ground fogging. Plexiglas was used because of ease of forming, but the heat generated by obtainable high speed will require the substitution of glass glass. The latter

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AVIATION WEEK, July 26, 1948

25

### Bell XS-1

#### Reaction Motors RML-6000-1 6000 Lb. Thrust

	Present Analysis (estimated)	Original Design (before pump)
type	25 ft.	25 ft.
Length	31 ft.	31 ft.
Height	11 ft.	11 ft.
Wing area	110 sq. ft.	110 sq. ft.
Empty weight	6937 lb.	6915 lb.
Gross weight	11,400 lb.	11,394 lb.
Maximum speed	31.3	31.3
	(5500 mph @ 55,000 ft.)	(5500 mph @ 55,000 ft.)
Maximum rate of climb	25,000 ftm	63,000 ftm
Maximum (700 ftm)	2.1 min.	4.2 min.
Landing speed	152 mph.	110 mph.

regimen special manufacturing techniques because of double curvature of the piece.

The coils are processed in a different order at 3.0 psi, and are tested to a load of no more than 1.0 psi before use. A control wheel is used, rather than a stick, to enable pilot to exert a moment of force if required. The rocket motor control system, instrumentation switches, and emergency controls are all located on the control wheel to enable pilot to operate them without moving his hands from the wheel.

► **Tanks—Power plant factors in the XS-1 consist of 5177 lb. of fuel and 210 lb. of solid waste loading to the dispersion of weight in a rocket aircraft.**

The liquid oxygen tank is located behind pilot, and lifts the rocket to the firing bank to the front edge of the wing. This tank is fabricated from 6161-T6 aluminum alloy steel and is designed to a burst 14 above the service pressure.

At this bank is the control chamber, which provides storage for the structural loading gear, the NACA instrumentation and two of the three engine nitrogen tanks on the engine frame. These tanks are made from 6061-T6 aluminum alloy NAK 9120 and are thickened from 0.175 to 0.625 in.

At the aft end of the tank is the ethyl alcohol tank of 0.175 in. nominal S&P 4131 steel.

The rocket motor is mounted in the extreme aft end of the airplane.

► **Fuel System Change—Delays in the development of several items of technology required to change in the XS-1 power plant.**

It was originally planned to provide pressure fuel for the propellant by a turbine-driven pump. This device represented a considerable development problem because its reliability had to be guaranteed in addition to its performance.

Long before the airplane entered construction a decision appeared that this unit could not be made in time, hence the fuel system was changed to a gas-actuated tank system. The propellant was forced to the aft chamber by the internal tank pressure rather than by the pump. This pressure is provided by nitrogen stored in the tanks previously described.

In this latter however, that area severely available of other nitrogen is sold as cylinders at only 2200 psi, far less than that required. This problem was solved by Bell's development of a nitrogen expander, in which liquid nitrogen is forced from a 5000 psi cylinder and split through a heat exchanger where the liquid "boils off" as high-pressure gas. The sphere surrounded has a wall thickness of 3 in., is cut in

halves, welded around the circumference, and will sustain a pressure of 3000 psi.

Gaseous nitrogen is piped from this expander into the nitrogen tanks within the plane. In addition to providing the propellant tank, this unit is also used to operate the landing gear, flaps, stabilizer actuator, and numerous valves. There is no control valve system on the airplane. Change in pressure fuel tanks is measured the empty weight of the XS-1 is about 2000 lb., reduced to 1500 lb. by about 1.5 inch. because of reduction in volume of fuel carried, and increased 210 lb. of solid waste loading to about 55 lb. per sq. ft.

The turbine-driven pumps have since been developed and will be installed in later models of the XS-1.

► **Power Plant Midspan**—The solid motor was developed by Reaction Motors, Inc., Pompano Lakes, N. J. It consists of two aqueous chambers grouped into a single motor. By using two small chambers rather than one large one, an approach to engine throttling was achieved, thereby increasing the pilot's control of thrust versus time.

The four chambers, each of which produce 1300 lb. thrust, can be fired separately through the switches on the control wheel. This provides variable thrust in 25, 50, 75 and 100 percent proportions. Ground plan for a variable load throttle on each chamber have not yet been completed.

The chamber operates at a pressure of 130 psi and its combustion temperature is about 3200 R. To cool the chambers, a regenerative system is used in which the fuel is circulated through a cooling jacket around each chamber before entering into the combustion chamber. The ethyl alcohol fuel is mixed with 25 percent of water to assist this cooling. The addition of the water has only a slight effect on the thrust of the motor.

Instrumentation of the XS-1 comprises its structural load. It is the transmission of these instruments into the fuselage support system that is the basic design problem. Of the three major questions posed by structural loads—stiffness and control, structural loads and pilot reactions—are answered by the special instrumentation.

► **Stiffness and Control Instruments**—These control control surface position, control operating forces, acceleration, rolling velocity, and rate slip rate. The control surface position is actuated by a slide wire transducer located on the rudders, elevator ailerons and horizontal stabilizer. As these surfaces are moved, the electrical resistance of the resistor is changed and recorded.

Control forces are obtained by strain gauges mounted on the rudder pedals, control column and control wheel. Changes in the resistance of these wires are recorded.

Acceleration of the airplane in the vertical, horizontal and longitudinal direction are obtained by a recording accelerometer, which consists of three flexible metal strips with weights at their ends, the weights leading to strain gauges in series through strain in the airplane structure to displace them.

Rolling velocity is recorded from a rate gyro, which indicates the rate of roll faster than this device of roll in the conventional gyro instrument.

Side slip angle is measured by a small vane located on a boom projecting from the port wing. As the side slip increases, the vane deflects into the wind and actuates a magnetic control transmitter.

To correlate these data, aircraft and attitude are recorded from static and dynamic pressure.

All recording is accomplished through a set of beams of light directed at a mirror and reflected to a set of film. This permits to be measured with the major portion of the aircraft, resulting in changes in the light source on the film. This film contains a time pulse trace and reference trace to permit interpretation and correlation of the data trace.

In addition to these recorded data, the data is elevator position, elevator position, airspeed and altitude are teleported to the cockpit through the flight log and is available in the event the airplane is damaged in the air or left in a crash.

► **Structural Loads**—These are measured on the structure mounted on the spine to the nose section of the wing and tail. These data are recorded in flight. Information on pressure distribution changes during the flight is obtained from 140 pressure orifices on the upper and lower surfaces of the port wing and 160 orifices on the tail, and data are recorded in flight.

As a check on the flight, ground tests are conducted to trace the path of the airplane. In flying the rate scope and indicating time intervals the speed and altitude of the craft may be determined at any given time during the flight.

This film is examined after the flight and coordinated with the attitude and airspeed recorded in the airplane.

► **Air Force, NACA Observers**—This instrumentation system is shared by the NACA over a period of months, some of the aerobically being in use, others being used in the XS-1 for the first time.

The Air Force airplane is less com-

pletely instrumented because of the different nature of the test program. In general, the Air Force objective is to determine maximum performance of the airplane both with respect to its speed, ceiling, rate of climb, etc. and structural strength. It is also to obtain from the XS-1 all data pertaining to design and operation of aerospace combat aircraft.

NACA objective is to obtain a maximum amount of aerobically instrumentation as aerodynamic aerodynamic phenomena.

For these reasons, the airplane was designed in cooperation (the Air Force and NACA) as a separate test program. In general, the NACA airplane 10-percent wing and 8-percent tail, is instrumentation and flight test program (the NACA craft is being flown at higher speed (maximum 5000) the speed of sound, the Air Force plane at much higher (maximum).

NACA research requires handle the instruments, their mounts and their displays, etc.

► **Roll Not Ended**—The attainment of optimum speed by the XS-1 airplane is only the beginning of its mission as a scientific research tool.

One of the major portions of the mission of this plane is the study of the various types of airflows, attitudes, through various maneuvers and under a wide range of atmospheric conditions to obtain data necessary to understand thoroughly the problems of the transonic zone.

For this reason the XS-1 will continue in service for several years and will be joined in the use of the aircraft by other types designed for the same purpose.

## British Plans for India

(McGraw-Hill World News)

**HYDERABAD**—Descent August, 1946, which the Indian's request withdrawal from Madras to Delhi up the country's "backbone"—giving British plans a wound.

Hyderabad has been the last of the British's last holdouts in India. Now an order has been placed in Britain for three Avrocar Aeroplanes.

The order reflects the trend in India Avrocar's a year ago every domestic aircraft and Dornier aircraft. The British plans are shuddering into an important place. Already in service are Vickers, Wyvern, and Doves.

Only two have been known to have Avrocar's been ordered to order. The Avrocar's, which are working for some General, and Indian Overcast Avrocar's (British's Metals Avrocar's) announced same time back it had ordered two Avrocar 210's.



THIS BLANKET of spun glass, substituted for graphite in an anti-friction agent, is also

below listed magnesium sheet being inserted from stretch press.

## Spun Glass Aids in Metal Forming

New material, woven into blankets, simplifies shaping magnesium sheets which must be worked when hot.

Spun glass, woven into this blanket, is being used at Glenn L. Martin Co., Baltimore, in the stretch forming of magnesium sheets as a successful substitute for less efficient materials.

In the stretch forming of magnesium sheets with light alloy, the stretch forming of sheet aluminum, it was found that the metal would not stretch properly because of friction set up by the junction between the sheet and the stretch block, causing an uneven flow which produced an unsatisfactory, irregular, wavy surface finish.

Further effort was made in industry for the solution of this problem in order to permit the material to be formed and the stretching block, as well, with a heavy coating of grease.

This liberal use of grease produces a messy and dangerous working condition. Then, after the stretching operation, removal of the grease frequently produces hard spots, and in addition, is a costly, time-consuming cleaning operation.

Substitution of a sheet of rubber for the grease offered favorable results for the cold forming of aluminum sheets.

► **Magnesium Different**—However, with the advent of magnesium in the sheet manufacturing field, the necessity for hot forming this material made the use of rubber unsatisfactory because it cracked and crumpled under the heat.

After considerable investigation by

Martin's research personnel, a 527-in. spun glass blanket was found to offer extremely favorable properties.

Effectiveness of the glass cloth was due to this application results from its best properties:

- Very high tensile strength when drawn tight.
- The hard smooth surface which the glass blanket films present to both work block and sheet metal, thus providing a sliding action between them.
- An accompanying property of the glass cloth which permits it to elongate partially with the sheet metal during the stretching operation because of natural flexibility.
- Practicability of stretching drawn heated sheet metal over an unheated block, since the glass blanket acts as a heat guard insulating, reducing transfer and loss of heat loss material to form block.

Use of the glass cloth at Martin indicates that it facilitates production, eliminates operations, and produces more satisfactory parts.

A representative test analysis of glass cloth is graphite as a friction reducing agent in stretch forming of a magnesium sheet 12½ x 6½ x .072 in. showing that operations are required by the graphite method, as for glass cloth.

Time required was 70.1 sec. for the graphite procedure as against 4.9 sec. for the glass cloth method.

## NEW AVIATION PRODUCTS



### Synchronous Engines

Synchronous differential, serving as intermediary synchronizing device for gear control equipment, is sponsored by **Kalifornia Instrument Div., Square D Co.**, 101-95 45 Ave., Elmhurst, N. Y. Single phase 440 volt 25 amp 17-47/64 in. long 14 in. dia. incorporating two synchronous motors and mechanical differential. Each motor reflects operating speed of engine and actuates 1/4 in. threaded output shaft rotating at half the difference of two motor speeds. When speeds are equalized output shaft becomes stationary. Efficiency and smooth operation are claimed through use of synchronous action. While motors do not require special excitation, it is stated they yield greater running torque than other self-synchronous types.



### Portable Photo-Developer

Photocopying and developing unit, **Copy-Rite Kit**, with complete processing facilities and self-contained dark room completely arranged in portable case, is offered by **Palisades Aerial Service, Inc.**, 13 West Union St., Poughkeepsie, N. Y. Unit is claimed quickly to reproduce prints, typewritten, drawings, or photographed material. Developing is by spray application. Material to be copied may be black or colored, single- or double-sided, lines or bands, transparent or opaque. Exposure time requires less than 1 in., fits into average size brief case or desk drawer, and can be used for copying without kit.



### For Filing Operations

Controlled corrosion descaling files, offered for slugging or controlling dirt in hand in **Blackford (EC)** without side involved in sanding and as hardening, are made by **Sevenson Tool Industries Inc.**, 546 Iron St., Sevenson, Mich. In addition to descaling, product is stated to be suitable for grinding, tapping, chamfers, and production pieces. Claimed it is used for production parts, output may be greatly increased. Available in four standard shapes: round, square, rectangular, and hexagonal. Produced chiefly for use in die-filing machines, units may be used separately or in any reciprocating device.

### Duo-Metal

New aluminum bearing alloy, **Ampco Grade 24**, for use in bearing and damping dies whose surface material must be free from scratches or whose galling would cause inadequate die life, is announced by **Ampco Metal, Inc.**, 101-101 45 Ave., Elmhurst, N. Y. Alloy is cast to specific patterns by sand or centrifugal methods, to provide die blocks that are extremely close to size, thus eliminating much preliminary machining.



### Gages Liquid Flow

Intended for measurement of fluid flow in external conduction systems, **Automatic Liquid Indicator**, made by **Engineering Research & Development Co.**, P. O. Box 166, Hurdle, Ill., is also adaptable to automatic flow determination of any liquid by weight. That measure is required for pure quantity of liquid to flow, as determined by balancing against calibrated weights, hence giving an independent of physical properties and temperature of fluid. Weighing section is installed in 24 x 18 x 9 in. cabinet with its movable glass door. Reservoir level is in fresh-sealed metal box. Device may be equipped with electric stop clock or timer.

### Removes Deposits

Applicable to almost overhead activities, **Carburetor, carburetor, cold test material for removal of carbon, sludge, grease and other** **Miles in Tests Products**, 6115 So. Central Ave., Los Angeles, Calif. Material consists of two layers, forming chemical seal and lower layer of cleaning agent.



### Pilot Set

Small pilot set No. 14, designed for aviation, marine, and industrial use, is available from **Electronic Tube & Tube Works, Allentown, Pa.** Set consists of three parts: first, a set of three tubes, and second, a set of three tubes, and third, a set of three tubes. The set is designed for use in a variety of applications, including aviation, marine, and industrial.

## FINANCIAL

### MARKET ACTION

Leading Aircraft Equities

Company	1944		1945		1946		1947		1948		1949		1950		1951		1952		1953		1954		1955		1956		1957		1958		1959		1960		1961		1962		1963		1964		1965		1966		1967		1968		1969		1970		1971		1972		1973		1974		1975		1976		1977		1978		1979		1980		1981		1982		1983		1984		1985		1986		1987		1988		1989		1990		1991		1992		1993		1994		1995		1996		1997		1998		1999		2000		2001		2002		2003		2004		2005		2006		2007		2008		2009		2010		2011		2012		2013		2014		2015		2016		2017		2018		2019		2020		2021		2022		2023		2024		2025		2026		2027		2028		2029		2030		2031		2032		2033		2034		2035		2036		2037		2038		2039		2040		2041		2042		2043		2044		2045		2046		2047		2048		2049		2050		2051		2052		2053		2054		2055		2056		2057		2058		2059		2060		2061		2062		2063		2064		2065		2066		2067		2068		2069		2070		2071		2072		2073		2074		2075		2076		2077		2078		2079		2080		2081		2082		2083		2084		2085		2086		2087		2088		2089		2090		2091		2092		2093		2094		2095		2096		2097		2098		2099		2100		2101		2102		2103		2104		2105		2106		2107		2108		2109		2110		2111		2112		2113		2114		2115		2116		2117		2118		2119		2120		2121		2122		2123		2124		2125		2126		2127		2128		2129		2130		2131		2132		2133		2134		2135		2136		2137		2138		2139		2140		2141		2142		2143		2144		2145		2146		2147		2148		2149		2150		2151		2152		2153		2154		2155		2156		2157		2158		2159		2160		2161		2162		2163		2164		2165		2166		2167		2168		2169		2170		2171		2172		2173		2174		2175		2176		2177		2178		2179		2180		2181		2182		2183		2184		2185		2186		2187		2188		2189		2190		2191		2192		2193		2194		2195		2196		2197		2198		2199		2200		2201		2202		2203		2204		2205		2206		2207		2208		2209		2210		2211		2212		2213		2214		2215		2216		2217		2218		2219		2220		2221		2222		2223		2224		2225		2226		2227		2228		2229		2230		2231		2232		2233		2234		2235		2236		2237		2238		2239		2240		2241		2242		2243		2244		2245		2246		2247		2248		2249		2250		2251		2252		2253		2254		2255		2256		2257		2258		2259		2260		2261		2262		2263		2264		2265		2266		2267		2268		2269		2270		2271		2272		2273		2274		2275		2276		2277		2278		2279		2280		2281		2282		2283		2284		2285		2286		2287		2288		2289		2290		2291		2292		2293		2294		2295		2296		2297		2298		2299		2300		2301		2302		2303		2304		2305		2306		2307		2308		2309		2310		2311		2312		2313		2314		2315		2316		2317		2318		2319		2320		2321		2322		2323		2324		2325		2326		2327		2328		2329		2330		2331		2332		2333		2334		2335		2336		2337		2338		2339		2340		2341		2342		2343		2344		2345		2346		2347		2348		2349		2350		2351		2352		2353		2354		2355		2356		2357		2358		2359		2360		2361		2362		2363		2364		2365		2366		2367		2368		2369		2370		2371		2372		2373		2374		2375		2376		2377		2378		2379		2380		2381		2382		2383		2384		2385		2386		2387		2388		2389		2390		2391		2392		2393		2394		2395		2396		2397		2398		2399		2400		2401		2402		2403		2404		2405		2406		2407		2408		2409		2410		2411		2412		2413		2414		2415		2416		2417		2418		2419		2420		2421		2422		2423		2424		2425		2426		2427		2428		2429		2430		2431		2432		2433		2434		2435		2436		2437		2438		2439		2440		2441		2442		2443		2444		2445		2446		2447		2448		2449		2450		2451		2452		2453		2454		2455		2456		2457		2458		2459		2460		2461		2462		2463		2464		2465		2466		2467		2468		2469		2470		2471		2472		2473		2474		2475		2476		2477		2478		2479		2480		2481		2482		2483		2484		2485		2486		2487		2488		2489		2490		2491		2492		2493		2494		2495		2496		2497		2498		2499		2500		2501		2502		2503		2504		2505		2506		2507		2508		2509		2510		2511		2512		2513		2514		2515		2516		2517		2518		2519		2520		2521		2522		2523		2524		2525		2526		2527		2528		2529		2530		2531		2532		2533		2534		2535		2536		2537		2538		2539		2540		2541		2542		2543		2544		2545		2546		2547		2548		2549		2550		2551		2552		2553		2554		2555		2556		2557		2558		2559		2560		2561		2562		2563		2564		2565		2566		2567		2568		2569		2570		2571		2572		2573		2574		2575		2576		2577		2578		2579		2580		2581		2582		2583		2584		2585		2586		2587		2588		2589		2590		2591		2592		2593		2594		2595		2596		2597		2598		2599		2600		2601		2602		2603		2604		2605		2606		2607		2608		2609		2610		2611		2612		2613		2614		2615		2616		2617		2618		2619		2620		2621		2622		2623		2624		2625		2626		2627		2628		2629		2630		2631		2632		2633		2634		2635		2636		2637		2638		2639		2640		2641		2642		2643		2644		2645		2646		2647		2648		2649		2650		2651		2652		2653		2654		2655		2656		2657		2658		2659		2660		2661		2662		2663		2664		2665		2666		2667		2668		2669		2670		2671		2672		2673		2674		2675		2676		2677		2678		2679		2680		2681		2682		2683		2684		2685		2686		2687		2688		2689		2690		2691		2692		2693		2694		2695		2696		2697		2698		2699		2700		2701		2702		2703		2704		2705		2706		2707		2708		2709		2710		2711		2712		2713		2714		2715		2716		2717		2718		2719		2720		2721		2722		2723		2724		2725		2726		2727		2728		2729		2730		2731		2732		2733		2734		2735		2736		2737		2738		2739		2740		2741		2742		2743		2744		2745		2746		2747		2748		2749		2750		2751		2752		2753		2754		2755		2756		2757		2758		2759		2760		2761		2762		2763		2764		2765		2766		2767		2768		2769		2770		2771		2772		2773		2774		2775		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Paris Letter

# French Feeble in Berlin Air Test

Reasons discussed in heated Assembly debate that results in industry changes; Air France affected.

By Boyd France

PARIS—Allied planes threatened into Berlin: even a minute but week bringing load and fuel to 2,000,000 pounds charged in the Red Army stronghold. About 500 in a French battleship tank 51, sporting the French tricolor on its tail, walked in for a landing. The French Army had been late to put into the air only its Junkers and not Daimler in the Wehrmacht had held Berlin. French ships took 34 hours scheduling together scored an average of 20 landings per day. Combined U.S. and British sorties passed the 400 mark daily.

► **Isotopes**—The French feeble fighters affected the outcome of the French aircraft industry battle. French Army aircraft were authorized by their pilots during explained that they are not yet built in India and North Africa.

But despite the French National Assembly level another storm. A Government in emergency told them that three years of civil aviation had given S.N.E.C.M.A. the state firm which refuel 50 percent of French airlines capacity. 510 million into the red, the end most of the aircraft turned out for the French Army would not run, that a whole consortium shipped to Poland was not built in studies.

► **Conservatism**—The Government showed the same squabbles as the Communists, who ran the Air Ministry and S.N.E.C.M.A. for nearly two years after the war. Communist demands sharply across three dimensions of having needed up S.N.E.C.M.A.: the Government at trying to swap out-of-control subsidies on order from Washington. Communist spokesmen they heard how the Communist administration had done a lot of getting on as well as bringing while they carried the firm. The debate ended for less than the Assembly, occasionally flying up into fighting, switches between Communist deputies and majority members. The Government came out of the tank with a run but declined to overhaul S.N.E.C.M.A. thoroughly.

more modern planes in 1946. They expect delivery that year of the Constellation 776, four C-54, 16 Languedoc, 10 DC-6, three de Havilland Doves, one Latécoere 611 flying boat. This would bring the total strength of their fleet up to 15 Constellations, 15 DC-6, four C-54, 16 Languedoc, 45 DC-3, one Junker 52, three de Havilland Doves, three Latécoere, and three Catalina.

A series of heavy maintenance workshops being equipped is expected to lighten the maintenance bill of the U.S. 52nd group, however, probably still continue to eat up only 40 percent of the airline's annual credit for new equipment.

The shortage of manpower which slowed down expansion in the period immediately following the war has been overcome. Air France's training school at Le Bourget has now housed 131 pilots, 71 mechanics, 215 technicians. It now, is expected to graduate an extra 1,000 men in 1948. Air France officials now believe that the French French personnel developing in the near future.

## Cut-Rate Fare Squabble

BOMBAY—An attempt to undercut passenger fares between India and Europe resulted in the destruction of a British Airtel Skyways' law for two days before the matter was straightened out.

The Saloon class, returning from Australia, had been booked on a charter basis by Indian Government Airlines to carry 40 passengers to India from Bombay. The fare was to be \$350 converted with the \$156 charged by Air India International and other international airlines.

An Indian International officials lodged a complaint with the Directorate of Civil Aviation claiming: (a) that regulations prohibited non-scheduled airlines from chartering scheduled airlines, and (b) that the fares violated IATA agreements.

After two days Indian Government officials pointed the difference in fares with the government, pending settlement of the question, and the plane was allowed to proceed.

## New Air Links for Italy

MILAN—Societa Italiana Servizi Aerei's (SISA) Trieste service connecting with French and Spanish air transport systems establishes a close link between the West and East.

Although the two airlines TWA's routes through Italy to the Levantine north, it represents a decided advantage for Italian air travelers—they'll be able

to pay for their trip with Italian currency. In addition to the Zen-Web network, SISA's newly established twice-weekly Trieste-Marseille service together with its regular Pagan-Milano route to Central and South America. Europe, on one side, and the Western Mediterranean on the other.

## British Permits

Charter operators get ministry approval for scheduled service.

LONDON—A shift in British aviation policy has put a dozen transport companies, formerly air charter operators, into scheduled service with government approval.

Since the war, the Civil Aviation Act had prevented such companies from establishing scheduled operations with any such regularly as might be then open to the charge of offering common carrier service. Although all air scheduled airlines within the British Isles was reserved to the nationalized companies, British European Airways, set up in 1946, was given a monopoly of all the scheduled international service.

Now, however, a number of British private operators have been granted permits by the Ministry of Civil Aviation to establish regular services between points within the United Kingdom over routes not now being served by BEA. (Some of the points previously were served by BEA or its predecessor.)

The new routes after representations by the British Air Charter Association

to the Ministry. In each case, the operators would call on BEA for the type and frequency of services, limit to be charged, etc. After BEA had approved these, the airlines were forwarded to the Ministry for its consent.

► **Schedule Reductions**—The private operators will run their respective services strictly to provide business visitors, they also will stand to suffer any loss or less in their operations. BEA will neither subsidize the deficits nor share in the profits.

Operators, of course, must conform to all the strict MCA and Air Registration Board standards that BEA observes.

The operators and routes to be served are as follows: (1) Eastern Airways, Western Airways, Manx-Warwick-Cardiff.

(2) Eastern Air Services, Cardiff-Cardiff-Winter.

(3) Olley Air Services, Cardiff-Cardiff-Newcastle, and Croydon-Croydon ( Isle of Wight ).

(4) British Air Transport, Croydon-Croydon-Gates ( Isle of Wight ).

(5) Air Enterprises, Croydon-Croydon ( Isle of Wight ), Croydon-Croydon ( Isle of Wight ), Birmingham-Liverpool, and Birmingham-Southampton.

(6) Lancashire Aircraft Corporation, London-London (Yorkshire Airport)-Isle of Man.

(7) Northern Air Services, Newcastle-Newcastle ( Isle of Wight ).

(8) Scottish Airlines, Perth-Perth-Blackpool, and Cardiff-Isle of Man.

(9) Moss Air Charter, Isle of Man-Isle of Man-Cardiff.

(10) Calais Air Navigation and Transport-Cardiff-Isle of Man.

(11) West Coastland Air Services-Cardiff-Isle of Man.

(Note: Because of the number of competitors also granted permits to operate between Cardiff and the Isle of Man, West Coastland Air Services may not take advantage of its permit.)

The permits are good for six months. ► **"Madding Through"**—Commenting on the new arrangements, "The Aeroplane" describes them as a "madding example of British 'madding through'." They are not believed to be either a device for obtaining competitive economic figures with which to check the efficiency of BEA as a means of proving that such services would not possibly pay their way.

For example, while BEA had been losing five at the rate of \$15 a day, far less serious on the Western-Cardiff route, Western Airways and Cardigan Air Services will be getting only \$3.14 a day for the same number of services under a new system of fares which come into effect on May 1. Further, the new operators are expected to be intended to get up the fares, increasing their loss for the West, one way and 16.4, second stage charged by BEA to 19.54 and 22.04 respectively. However, if the independent operators can make this service pay, there will be no loss of competition with BEA's effort.

It is understood, however, that while the permits are granted for only six months, the intention is to continue the permission and not merely to let it slip during the central investigation period a number of private operators who would then be offered the choice of going out of business or being absorbed by BEA.



BREGUET 420—360-ENGINE AND ALL

Swiss Breguet 420s built for the Swiss Air Force, the prototype of the four-engine transport, the Breguet 420, started in 1946 (Aviation Week, June 21). Outfitted with four British Bristol Perseus propellers,

the plane is a low-wing structure, with cockpit leading gear and high-winged nacelles. Various versions of the plane measured 55 ft, 56 ft, 64 ft, and 66 ft long; gear 36 ft, 38 ft, 40 ft, and 42 ft; and

flight wingspan, and a design carrying twelve berths and 17 chairs. Plane has overall length of 130 ft, with 131 ft, 10 in. wing. Cruising speed is 264 mph, maximum speed 344 mph. Cockpit height is 46 ft.







## Some Short jettings for airline operators and their crews



World routes in the air and heavy duty aircraft

## What does the modern business man want . . . ?

### Who says he's in a hurry?

As air travel becomes more and more an everyday affair the popular press assumes one nothing event after another for modern times between various travellers. In order to meet the importance of these facts, the press has had also to produce someone who really wanted to move about the world as these modern speeds, and who lived up to his better than the "modern business man". But, with vast demands on his time in all parts of the world, would appear to be just the man who would be prepared to sacrifice any amount of comfort for speed.

### Has he asked for speed?

Experienced operators of some of the world's largest and best-known airlines have dealt with every type of passenger over many years. They feel that if this type of business man comes in, it will be in such a way as to be almost legendary. They have often, and quite different, ideas of what the modern

business man requires on his long or short journey, by air. He wants to avoid no time, travel in comfort, and have no pleasure in a break from business routine as possible.

### How to get him there as fast . . .

No time is wasted where schedules can be quickly kept and this can be achieved more easily, no suggest, with a fleet of flying boats than with any other type of aircraft. Business men—and any other passengers for that matter—must know that they will arrive at a given place in a given time and be able to work their appointments to match. If they also have time at the momentaneous stage either to enjoy themselves or to other business, so much the better.



### How to make him comfortable

Give him plenty of room. Don't cramp him. Give him a cabin with not more than about five fellow passengers whose faces he can see—he doesn't want to watch the backs of their heads, or the backs of their chairs, possibly for days. Give him large windows with a wide view of the surrounding country. He'll get all done and a separate promenade cabin and cocktail bar—room to move about and a changing atmosphere and view—also there flying boat. Then he'll find it's fun to fly by flying boat and it's a relief that much.



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## AIR TRANSPORT

### Costs Pushing Fares Up Again

United files tariff raising passenger rates by one-half cent a mile in what may be prelude to general boost.

The domestic airlines may be losing market on land with inflation. United Air Lines has set the wheels in motion for what may be the third industry-wide 10 percent passenger fare increase of the postwar period. A new tariff boosting UAL's rates from 3.5 to 4.0 cents a mile effective Sept. 1 was to be filed with CAB last week.

Other Carriers Move—The move brought no immediate reaction from other carriers, although sentiment for the fare hike has been developing. Even before United's announcement, Northwest Airlines President Carl Hunter had indicated his company would act as increase this fall.

W. A. Parnass, UAL president, and the new fare boost was made necessary when increased efficiency failed to offset the effect of steadily rising costs. "We expect this fare increase to help," he declared, "but whether it is sufficient will depend on future cost trends and road conditions as well as just what United techs are successful."

Basic air way domestic airline fares today are on an upward trend since 1949 despite the continued inflation but remain pointed out. "In contrast, one

way Pullman fares were approximately 37 percent higher than prewar levels even prior to the late-war Commerce Commission's recent action granting on other passenger rate increase averaging 17 percent to eastern railroads. And the general index price level has risen to 177 percent of 1941."

Traffic Slump a Factor—Like most other carriers, United's need for more revenue stemmed in part from the passenger traffic slump during the first five months of this year. In June, for the first time in 1949, UAL's revenue passenger mileage was above that of the same month last year, showing a gain of over 14 percent.

United's move toward higher fares followed closely upon a CAB ruling that it would not use and pay to compensate a carrier for all the increased costs arising from the current inflation spiral. In its Chicago to Southern Air Lines rate opinion (Airlines Week, July 19), the Board said "Further increases in passenger and cargo rates may be required if rising price and wage levels continue to be reflected as increased cost of revenue traffic."

Three Boosts for NWA—Northwest

Airlines, while leaving the new fare level, was in a more difficult position than other carriers with regard to lowering the movement. It has submitted three bills since March, 1947, against two for the remainder of the industry.

Last October, NWA made effective the first semi-annual passenger fare hike. Then, on May 15 of this year, it lifted rates another eight percent. Effect of Northwest's third-round hike was swamped by its 10 percent raised trip discount and the plan for reducing five percent of the ticket price to passengers on planes arriving at their destinations after 10 minutes. (Airlines Week, Feb. 10). Early this month NWA indicated to CAB it could not seek higher passenger rates until certain had met its current fuel bill.

Failure of other carriers to follow United immediately in seeking higher fares presents a painful wait for the industry. Northwest took the lead of their loss, seeking its income offset in October. But the remainder of the industry was slow to follow suit. American not falling in line said De Cosmides, and Eastern and National delaying action until January.

NWA "On-Time" Program Northwest Airlines' "on-time" program is paying off. Since its inception last March, 98.6 percent of all flights have been ahead of the 30-minute deadline, Carl Hunter, NWA president, recently announced. This is all part of the airline's cash-backed intention to keep its flights on schedule. A 5 percent refund is made to any passenger whose arrival is delayed more than 30 min.



### PAN AM STRATOBOEISER

The American World Airways new Clipper America, first of its order of 10 Boeing Stratobuses to be delivered in November.

York-London flight time is twelve hours. Each carries 75 passengers and has a lounge and bar on the lower of its two decks.

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## Use of Flight Engineers Protested

American leads battle against ruling, claiming third crew member on DC-6 may be safety hazard.

The airlines still are battling the new Civil Air Regulations which require use of a flight engineer-in addition to the pilot and copilot-on DC-6s, Boeing Stratocruisers and other aircraft certified for more than 50,000 lb. takeoff weight.

American Airlines, sponsor of the largest fleet of DC-6s (78), is leading the fight on the rules announced by CAA in April (Aviation Week, Apr. 26) to take effect Dec. 1. In a petition for relief, AA asserted that full inspection of a flight engineer would constitute a definite hazard to safety.

**Accident Records Cited**—"Our conviction is that rapid," American declared, "is borne out by the history and record of the DC-6 cockpit, by the absence of any doubts which could be delegated to an engineer, and by the accident record of non-engineer aircraft with and without engineers."

AA maintains there should be no blanket Civil Air Regulations requiring employment of an engineer on aircraft in excess of 50,000 lb., or any other specific weight. Besides being illogical, the carrier contended, such regulations will restrict severely the activities of non-operators and carriers in their attempts to design engine cockpits of ever increasing simplicity. "This would and

hinders the progressive evolution of the simplified two-man cockpit—use of the more safety features of the DC-6."

**Research Required**—American said that questions of crew composition are technical problems which should be decided only after scientific research by manufacturers, carriers and the Civil Aeronautics Administration.

The petition stated that American is concerned over the absence of basic scientific information and research with respect to two of the main problems of an engineer safety (1) simplification of cockpit design, and (2) coordination between pilot and copilot, or between the members of a multiple crew.

**Investigation Urged**—"We believe CAA should conduct a thorough investigation, in conjunction with the Civil Aeronautics Administration, the Air Force, the Navy and certain private organizations, into all such problems," AA declared. "Pending the results of such an investigation, it should not take the dangerous step of requiring a third crew member on a two-man cockpit with or without the DC-6."

Asserting "there is nothing whatever for a flight engineer to do in a DC-6," American explained with charts and statistics why operation of the plane with the extra crewman is a hazard. It

emphasized that the DC-6 cockpit is currently the most recent DC-4 cockpit with the addition of newly developed means of easing the pilots' load, adding that with such a setup a flight engineer could only "get in the way."

**Functions Limited**—AA said historically the flight engineer has performed one or more of the following functions:

- Inspection and servicing of planes on the ground or in flight, or inadequately trained mechanics—"This was the function of wartime flight engineers in case of crash. Obviously there is no need for such a man in American's DC-6 operation, where qualified ground crew provide necessary service at each station."

- Repair in flight—"Some few aircraft, such as the Boeing 314 flying boats, were built to permit access to engines in flight. In an airplane so constructed, the flight engineer has possible duties. The DC-6 is not such an airplane."

- Fuel conservation—"This has been the principal function of flight engineers involved in long-haul or cruise operations. The joint efforts of airlines and aircraft engine and instrument manufacturers have cost so many millions of dollars in fuel conservation, automatic carburetors, liquid fuelers for direct reading of engine power fuel flow meters, etc. With this evolution of direct and more simple techniques, the importance of the flight engineer with his 'long-gone' career is fast disappearing even on transoceanic routes."

- Engine power control—"In some aircraft, by means of design concepts or construction, it is impossible to locate the required controls and instruments within visual and manual range of the pilots. In such cases, it becomes necessary that certain selected instruments and controls, frequently those affecting the control of engine power, be separated and mounted to another panel which then necessarily becomes the status of the flight engineer's panel. The DC-6 is not such a plane. In conception, construction and certification it is a two-man aircraft."

- Safety Threat Seen—American said that actual experience with heavy planes shows that a flight engineer does not add to safety. "Actually the opposite is true. There have been roughly twice as many accidents involving fatalities per mile of non-engineer plane miles flown as with flight engineers on without them."

"Some accidents can be traced directly to flight engineer error as a result of contributing agent, whereas an accident has been traced to the absence of a flight engineer." No DC-6 accident has been traced to the absence of a flight engineer or to an excessive work load on the pilots."

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**SAFETY COUNCIL AWARD**

Mid-Continent Airlines President J. W. Nelson (center) in class receiving the National Safety Council's 1957 Aviation Award from Albert H. Wood (center), retiring head of the Kansas City Safety Council, who made the presentation in behalf of Sgt. H. Decker, president of the National Safety Council. The plane (left) Mid-Continent for flying 282,622,000 passenger miles in 15 years without a passenger or crew fatality. Lane W. Crowl (left) was cited in Nelson Wood.





To make these extra heat protection Delatubeston aircraft were not made to meet any existing heat protection standard. These aircraft are built to a much higher standard than most aircraft, including those that are built to meet the requirements of the Federal Aviation Administration (FAA). The aircraft are built to a much higher standard than most aircraft, including those that are built to meet the requirements of the Federal Aviation Administration (FAA). The aircraft are built to a much higher standard than most aircraft, including those that are built to meet the requirements of the Federal Aviation Administration (FAA).

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**GENERAL ELECTRIC**

## CAB Broadens Attack on Nonskeds

More operators are added to list of nonskeds accused of violating the nonscheduled exemption.

The Civil Aeronautics Board is stepping up its campaign against alleged violators of its nonscheduled exemption.

Two operators on the U.S. Alaska route are the latest to become entangled in CAB enforcement proceedings. They are Mr. McKinley Airways, Anchorage, Alaska, which has been using two DC-1s, and Golden North Airways, Fairbanks, Alaska, which flies two DC-1s and a C-47. Both carriers were ordered to show cause by Aug. 9 why they should not be forced to cease and desist from violating the Civil Aeronautics Act.

Flight Two Frequent-CAB action, however, charged that although Mr. McKinley has authority to operate only as an "operator," it had in fact made regular and frequent flights between Seattle and Anchorage. The carrier's U.S. Alaska trips originated between Seattle and Fairbanks monthly from October, 1947, through March, 1948.

Golden North was cited for operating frequent and regular trips primarily between Seattle and Fairbanks. CAB said the irregular carrier made between eight and 15 flights per month from October 1947, through March, 1948.

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## Seaboard & Western Shows \$11,209 Profit

Seaboard & Western Airlines, Inc., which has made a profit since it started flying, had an earned income, before taxes, of \$11,209 for the seven-month period ending May 31, 1948.

In report to the Civil Aeronautics Board, filed last week under new CAB regulations affecting irregular air service, stated that at the end of its first fiscal year Aug. 31, 1947, earned income was \$40,834 before taxes.

► No Annual Pay—President Raymond A. Naden, explaining that the earnings were achieved without benefit of annual pay, and his company had flown 2,618,885 freight ton miles since it started flying operations at the close of May, 1947, through May, 1948.

In its peak month last February, Seaboard flew 421,759 freight ton miles. Thirty-nine of the 157 shipments on outbound flights exceeded 100 lb each and of those 20 were between 3000 and 2000 lb, and three exceeded 12,000 lb. Average shipment was 857 lb.

Development of international air freight, he said, points to an "immense potential."

► Flight Month Population—Organized in 1946, Seaboard began its first scheduled passenger service to Europe and the Middle East, negotiating for landing rights, and offering and running flight crews. Flying began with one DC-4, converted for all-weather operation. Four have been added.

Naden reported that during the carrier's first operational year its civil cost paid 224 flights over the North Atlantic and beyond more than 5,000 flight hours.

► Varied Cargo—Shipments of varying weight constituted 46 percent of the total business, though cargo varied widely. The company has carried some passengers on a contractual basis.

Disputed one of the contracted airlines supporting the Military Air Transport Service until, to Berlin, Seaboard flew the first flight in the last 15 months of July 1 and was ordered to do so at the 15 up by the Air Force last week to carry 100 refugees for the operation. Other in the report is concerned with Transwestern Airlines, four, Alaska Airlines, Inc., and the American Airways, one.

► Puerto Rican Service—Then enforcement proceedings are pending against American Airlines, operating between the U.S. and Puerto Rico. American Air Transport and Flight Schools, Inc., Miami Springs, Fla., was ordered to cease flying by July 30 why it should be in the order to cease and desist from violating the Civil Aeronautics Act. The company, which has two DC-4s, allegedly operated with two great irregularities and regularly between New York, Miami and San Juan, and it was found to charge the fare specified in its tariffs.

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## Airplane to Begin Service

American Airways, Phoenix, plans to inaugurate service on its semi-regular scheduled freight service early in August with two DC-3s connecting daily between Phoenix, Prescott, Flagstaff and Winslow, Ariz. Two weeks after starting service, Arizona Airways hopes to extend operations to other points.

**Ryan Navion**

RYAN AERONAUTICAL COMPANY, 400 Lindbergh Field, San Diego, California









# EDITORIAL

## Two Questions for Personal Plane Makers

In the short period since publication by Harvard University of the book "Personal Aircraft Business at Airports," a spot check among fixed base operators and personal airplane owners finds many who agree with the diagnosis of Prof. Lynn Bolinger and Arthur Tully, the authors.

They denied that the main trouble with the selling personal aircraft business was the product it had to sell. They recommended that a design with ease of sale be offered so that the market would grow. Such a design should be a plane capable of using very small fields, which could be placed inconspicuously near centers of population. This would require the airplane to have shortness, plus short takeoff and landing characteristics.

In all fairness, a number of airplanes now on the market have a real utility which is not sufficiently appreciated by anyone except the small and slowly growing group of people who own and use them. Particularly useful are the homeplace airplanes of relatively high performance whose cross-country time approaches and in some cases surpasses two-engine airline time for the same trips.

However, these airplanes have not been designed for the very small close-in landing facilities that Bolinger and Tully talk about. Their landing speed and their take off run are still in such large figures that only a highly skilled demonstrator pilot could put most of them safely into the "ball-park" sized landing facilities called for.

It would take on design muscle to provide a similar airplane which could operate out of very small fields but at sacrifices in cruising speed, power and initial low cost.

Besides the conventional fixed wing airplane with high lift devices for small field performance, there are other possible answers to this problem. Some of the more obvious:

- The helicopter.
- A convertiplane which succeeds and decelerates like a helicopter but levels off for cruising flight like an airplane.
- An aircraft using some form of boundary layer control such as, possibly, the experimental Coe's channel wing.

Up until now, the rank-and-file American lightplane company has declined to make an airplane designed to use the existing airport pattern. We know of no airplane being marketed which can run safely and comfortably on an airstrip 200 by 600 ft.

Federal technical and safety regulations also are a major factor in any plane manufacturer's decision to make the drastic change in his product that Bolinger and Tully want.

Experience with CAA legislation has not been such as to encourage the manufacturer to take such a step. How

over the manufacturer are now studying the definite advice of CAA, made months ago by former Administrator T. P. Wright, to give them much more freedom in certification of aircraft.

Sensible recognition of the responsibility that goes with such freedom has caused the manufacturers to take a second look at the whole matter.

Yet, a serious overhauling of CAA technical regulations governing personal aircraft would be a very important and helpful step in the recommended product change. A definite understanding that slow flying aircraft would get special recognition in safety regulations would be another helpful step.

Technically the helicopter has such recognition now. Actually the red tape and created in the worst interpretation by CAA is a Washington, D. C. helicopter case has thrown a serious cloud over the whole helicopter operation situation.

If there should be made available a four place personal and business plane with the characteristics described, it would be a relatively inexpensive matter to provide close in landing facilities in most cities. The helicopter airports which are being developed at Chicago, Milwaukee and Cleveland for today's airplanes are far more expensive because of their size than such a miniature airstrip would be.

Besides increased utility, the increased safety factor of a relatively lightly loaded aircraft with slow landing speed and short takeoff needs serious consideration. The Civil Aeronautics Research Directorate has made studies pointing to the fact that built in crash protection is man in after crashes as the speed of the plane drops. This actually is only a substantiation of a basic fact often demonstrated in auto crashes.

Investigators now going on with a slow flying fixed-wing airplane at Aeronautical Research, Inc., Boston, indicate that a plane with a maneuverable speed of 40 mph and crash protection effective at 50 mph, is in sight experimentally.

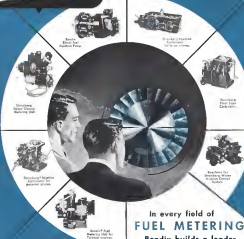
Admittedly, the first cost of such airplanes would be higher than that of current models. But two obvious questions present themselves:

- Will not consumers be willing to pay a higher cost to get an aircraft with such increased safety and utility?
- Will not the higher consumer demand created by such increased utility and safety result in greater volume of production than today's with attendant lowering of unit cost?

(Editor's Note: In the vacuous absence of Robert H. Wood, Editor of *Aerobics* News, this editorial has been contributed by Alexander McQuade, associate editor.)

# Bendix Products

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## DESIGNERS AND BUILDERS OF THE WORLD'S FINEST A I R C R A F T

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